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简报

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上单翼飞机起落架整流罩减阻研究

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Drag reduction investigation on high wing aircraft landing gear fairing

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- 摘要
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摘要 上单翼飞机起落架整流罩一般体积较大,需要设计良好的气动外形以尽可能减小阻力。为明确起落架整流罩的气动特性影响及指 导修形,首先进行了有无起落架整流罩的气动特性对比分析。然后基于CATIA的NURBS参数化方法,考虑起落架布局约束,对起落架整 流罩进行了修形设计。CFD计算结果表明,修形后总阻力减小1.9个Counts,约占27%的减阻空间。另外为明确整流罩的影响,释放了 起落架约束进行修形,阻力减小可达到3.4个Counts,并进一步确定整流罩的影响主要体现在对机翼的干扰上。

关键词: 起落架整流罩 修形设计 NURBS 参数化设计

Abstract: High-wing aircraft generally has a big landing gear fairing, thereby its shape needs to be designed carefully so as to reduce drag as much as possible. In order to understand the aerodynamic effect of landing gear fairing and guide the following fairing shape redesign, a contrast is made between the configurations with fairing and the otler without fairing. Considering the landing gear layout restraints, a NURBS parametric modeling method within CATIA is proposed and implemented to redesign the shape of landing gear fairing. CFD results indicate that the drag is reduced by 1.9 Counts, about 27% of the whole drag reduction margin. Moreover, in order to understand the fairing effect more clearly, a deformation without landing gear constraints is performed. The drag reduction can reach to 3.4 Counts, and results show the effect of fairing can be concluded as the disturbance between fairing and wing.

Keywords: landinggearfairing, shaperedesign, NURBS, parametricdesign

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