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药柱含扰流板 $\mathrm{H_2O_2}/\mathrm{HTPB}$ 固液火箭发动机 两相流数值计算

Two-phase flow numerical calculation of ${\rm H_2O_2/HTPB}$ hybrid rocket engine with diaphragm in grain

投稿时间: 2012-10-26

DOI:

中文关键词: 固液火箭发动机 扰流板 药柱 数值计算 两相流

英文关键词:hybrid rocket engine diaphragm grain numerical calculation two-phase flow

基金项目:

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中文摘要:

The effect of diaphragm in grain on combustion performence of hybrid rocket engine was studied. The standard ϕ 100mm hybrid rocket engine of Beijing University of Aeronautics and Astronautics was selected as a computational model, and 98% hydrogen peroxide (H_2O_2)/hydroxyl terminated polybutadiene (HTPB) was combined as propellant; different hole diameters of diaphragm setting at different grain positions were computed by 2-D axial symmetry and gas-liquid two-phase flow model. The results show that smaller diameter of diaphragm hole and the diaphragm setting at positions ranging from 50% to 70% of grain length can obtain the largest averaged regression rate of fuel and the highest combustion efficiency. This conclusion will provide a theory evidence for design of the diaphragm.

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