



航空学报 » 1994, Vol. 15 » Issue (4) : 433-439 DOI:

论文

[最新目录](#) | [下期目录](#) | [过刊浏览](#) | [高级检索](#)

[<<](#) [<](#) [前一篇](#) | [后一篇](#) [>](#) [>>](#)

应用能量—Casimir方法研究充液卫星系统的运动稳定性

匡金炉, 黄克累

北京航空航天大学应用数理系, 北京, 100083

THE NONLINEAR STABILITY OF THE LIQUID-FILLED SATELLITE WITH FOUR-FLEXIBLE-ATTACHMENTS USING THE ENERGY-CASIMIR METHOD

Kuang Jinlu, Huang Kelei

Dept. of Appl. Math. and Mech., Beijing University of Aeronautics and Astronautics, Beijing, 100083

摘要

参考文献

相关文章

Download: [PDF \(385KB\)](#) [HTML](#) 0KB Export: [BibTeX](#) or [EndNote \(RIS\)](#) [Supporting Info](#)

摘要 利用能量—Casimir方法研究了带4个挠性梁的充液卫星系统在外力矩状态下,关于绕铅垂轴稳态转动的非线性稳定性条件,该条件考虑了液体的涡旋、弹性梁的振动、卫星主刚体的旋转以及流—弹—刚之间的耦合,此外还考虑了离心力与Coriolis惯性力的影响。推广了Rumjantsev和Marsden的部分结果,为带挠性梁的充液航天器的运动稳定性总体设计提供了可靠的理论依据。

关键词: 充液卫星 柔性附件 能量—Casimir方法

Abstract: Nonlinear self-spinning stability criteria of liquid-filled satellite with four-flexible-appendages are investigated with the Energy-Casimir method. The stability conditions take into account the vorticity of the liquid, the vibration of the flexible appendages, rotations of the rigid satellite, the coupling of liquid-elasticity-rigid, and centrifugal and coriolis forces. These results which are the extension of Rumjantsev's and Marsden's are obtained for the first time.

Keywords: liquid-filled satellite flexible appendages Energy-Casimir method

Received 1993-06-14; published 1994-04-25

引用本文:

匡金炉;黄克累. 应用能量—Casimir方法研究充液卫星系统的运动稳定性[J]. 航空学报, 1994, 15(4): 433-439.

Kuang Jinlu; Huang Kelei. THE NONLINEAR STABILITY OF THE LIQUID-FILLED SATELLITE WITH FOUR-FLEXIBLE-ATTACHMENTS USING THE ENERGY-CASIMIR METHOD[J]. Acta Aeronautica et Astronautica Sinica, 1994, 15(4): 433-439.

Service

- ▶ [把本文推荐给朋友](#)
- ▶ [加入我的书架](#)
- ▶ [加入引用管理器](#)
- ▶ [Email Alert](#)
- ▶ [RSS](#)

作者相关文章

- ▶ [匡金炉](#)
- ▶ [黄克累](#)