首页 | 关于本刊 | 编 委 会 | 最新录用 | 过刊浏览 | 期刊征订 | 下载中心 | 广告服务 | 博客 | 论坛 | 联系我们 | English

















航空学报 » 2013, Vol. 34 » Issue (3):517-524 DOI: 10.7527/S1000-6893.2013.0087

流体力学与飞行力学

最新目录 | 下期目录 | 过刊浏览 | 高级检索

< ◀◀ 前一篇

后一篇 >



发动机进气道短舱前缘结冰三维模拟研究

申晓斌1,2, 林贵平1,2, 卜雪琴1,2, 郁嘉1, 侯盼雪1,2

- 1. 北京航空航天大学 航空科学与工程学院, 北京 100191;
- 2. 北京航空航天大学 人机工效与环境控制重点学科实验室, 北京 100191

Three-dimensional Simulation Research on Ice Shape at Engine Inlet Nacelle Front

SHEN Xiaobin^{1,2}, LIN Guiping^{1,2}, BU Xueqin^{1,2}, YU Jia¹, HOU Panxue^{1,2}

- 1. School of Aeronautic Science and Engineering, Beihang University, Beijing 100191, China;
- 2. Laboratory of Fundamental Science on Ergonomics and Environment Control, Beihang University, Beijing 100191, China

摘要

参考文献

相关文章

Download: PDF (3588KB) HTML KB Export: BibTeX or EndNote (RIS) Supporting Info

摘要

为计算发动机进气道短舱前缘的结冰冰形,在Messinger结冰热力学模型的基础上发展了一套纯三维的表面溢流水流动结冰模型。考虑空气摩擦力为表面溢流水流动的主要驱动力,用空气对进气道表面的剪切力来确定溢流水的流动方向及流量分配。为求解溢流水结冰模型,发展了一套重复查找表面控制体状态的方法,能够快速完成整个三维表面的计算。用该方法对某三维发动机进气道进行计算,得到了三维结冰冰形,并将计算结果与FENSAP-ICE计算结果进行了对比,结果显示两者的冰形轮廓基本一致,仅在冰角处存在差异,表明本文三维发动机结冰计算模型与计算方法是有效的,其计算精度与FENSAP-ICE结果相当。

关键词: 发动机 三维模拟 冰形 溢流水 热力学模型

Abstract:

To simulate the ice shape at an engine inlet nacelle front, a pure three-dimensional ice accretion model based on Messinger's thermodynamic model is established, which took the surface runback water stream into consideration. Regarding the air friction as the main factor to drive the surface runback water stream, the shear force of the inlet surface is used to determine the direction and flow distribution of the runback water. To solve the ice accretion model of the water film, an approach of repeatedly searching the surface control volume condition is developed, which could achieve the calculation of the whole three-dimensional surface. Ice accretion calculation of a certain three-dimensional engine inlet is carried out, and the three-dimensional ice shapes are obtained. Comparison between the computational results and FESAP-ICE results indicates that the ice profiles of both methods are coincident approximately except for the difference at the ice horn location. The comparison results confirm the validity of the three-dimensional engine ice accretion model and the computational method, and the precision of the results is almost the same as that of the FENSAP-ICE software.

Keywords: engine three-dimensional simulation ice shape runback water thermodynamic model

Received 2012-04-09;

Fund:

"凡舟"青年科研基金(20110503)

Corresponding Authors: Tel.: 010-82338600 E-mail: buxueqin@buaa.edu.cn Email: buxueqin@buaa.edu.cn

About author: 申晓斌 男, 博士研究生。主要研究方向: 流体力学与飞机结冰、 防除冰技术。 Tel: 010-82338469 E-mail: sxb762@163.com; 林贵平 男, 博士, 教授, 博士生导师。主要研究方向: 环境控制、 飞机防除冰及强化传热。 Tel: 010-82327533 E-mail: gplin@buaa.edu.cn; 卜雪琴 女, 博士, 讲师, 硕士生导师。主要研究方向: 飞机防除冰技术, 飞行器环境控制, 流动与传热。 Tel: 010-

Service

- ▶ 把本文推荐给朋友
- ▶ 加入我的书架
- ▶ 加入引用管理器
- ▶ Email Alert
- **▶** RSS

作者相关文章

- ▶ 申晓斌
- ▶ 林贵平
- ト雪琴
- ▶郁嘉
- ▶ 侯盼雪

82338600 E-mail: buxueqin@buaa.edu.cn

引用本文:

申晓斌, 林贵平, 卜雪琴, 郁嘉, 侯盼雪. 发动机进气道短舱前缘结冰三维模拟研究[J]. 航空学报, 2013, 34(3): 517-524.DOI: 10.7527/S1000-6893.2013.0087

SHEN Xiaobin, LIN Guiping, BU Xueqin, YU Jia, HOU Panxue. Three-dimensional Simulation Research on Ice Shape at Engine Inlet Nacelle Front[J]. Acta Aeronautica et Astronautica Sinica, 2013, 34(3): 517-524.DOI: 10.7527/S1000-6893.2013.0087

Copyright 2010 by 航空学报