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翼-身组合体绕流的Euler方程数值模拟

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NUMERICAL SIMULATION OF THE FLOW PAST WING-BODY USING EULER EQUATION

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摘要 参考文献 相关文章

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摘要用代数方法生成翼-身组合体H-O型网格,并用有限体积法研制出翼-身组合体绕流三维Euler方程计算程序。该方法的特点是改进了机身与机翼表面网格点分布,机翼后缘有后掠时也能保证后缘与网格线一致。程序除能按常规提供横流截面展向压强分布外,还能提供弦向压强分布。对 NASA TND-712的翼-身组合体模型的计算结果与实验符合很好。

关键词: Euler方程 有限体积法 冀-身组合体 数值计算

Abstract: This paper gives H-O type grid generation and applies finite-volume method for wing-body flow. A 3-D Euler code for wing-body has been developed. We improve in this code the distribution of the number of the grid points over wing and body surfaces, and keep the trailing edge of the wing being one of the grid lines even when the edge has sweepback or sweep forward. The code developed can provide not only the spanwise pressure distribution as usual on cross-flow planes, but also the chordwise pressure distributions. The computation for NASA TN D-712 wing-body modd with the present code shows that the computed pressure distributions are in very good agreement with the experiment.

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Keywords: Euler equation finite-volume method wing-body combination computation numerical

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